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ON THE CHOICE OF THE BALLISTIC PARAMETERS OF AN ON-ORBIT SERVICE SPACECRAFT

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At present, a significant increase in the cost of spacecraft is observed. Due to this fact the requirements for their active life duration, operational reliability, and operational cost reduction become more and more stringent. A promising way to meet these requirements is the introduction of on-orbit service (OOS). OOS allows one to solve technical and economic problems by performing service operations in space. The introduction of OOS contributes to extending the active life of spacecraft, increasing their operational reliability, and reducing the service maintenance cost of orbital satellite systems of various purposes. OOS programs on the deorbit of used or damaged spacecraft help in the mitigation of space debris problem. The composition of an OOS system depends on the service tasks to be performed and the ballistic capabilities of disposable or reusable service spacecraft. The realizability and character of the ballistic maneuvers of service spacecraft are largely determined by the type and characteristics of their sustainer engines. The aim of this paper is to assess the ballistic potential of modern and prospective OOS spacecraft and to develop a methodology for planning rational OOS routes. Various ground- and space-based OOS systems are considered and analyzed. The expediency of their use is estimated depending on the service tasks to be performed. The most promising OOS schemes are identified. A technique for planning a rational sequence of orbit transfers between the orbits of the spacecraft to be serviced is proposed and illustrated by the example of a test calculation. The technique is based on the solution of a multi-criteria traveling salesman problem, which is formulated in terms of integer linear programming and reduced to a single-criterion problem by the additive convolution method. The novelty of the proposed technique lies in reducing the original problem to a multi-criteria traveling salesman problem. The results obtained may be used in the justification, planning, and implementation of service space operations.

Keywords: traveling salesman problem, spacecraft, multi-criteria optimization, on-orbit service, route planning.

- 1 *Ivanov V. M.* Conceptual foundations of orbital servicing for advanced automatic space vehicles (*in Russian*). Spacecraft and Rockets. Vestnik Moskovskogo Aviatsionnogo Instituta. 2008. V. 15. No. 3. Pp. 5 7.
- 2 Barbee B. Design of Spacecraft Missions to Remove Multiple Orbital Debris Objects. Specialist Conference, Paper AAS 15-598, August 2015.
- 3 Stephen J. Design for on-orbit spacecraft servicing. Specialist Conference, Paper AAS 14-374, October 2014.
- 4 Caveny L. Space Engines: State of the Art and Prospects (in Russian). Moscow : Mir, 1988. 454 pp.
- 5 Serdyuk V. K., Tolyarenko N. V., Khlebnikov N. N. Space program support vehicles (in Russian). Rocket Production and Space Engineering: VINITI. Moscow. 1990. V. 11.
- 6 Gusev Yu. G., Pilnikov A. V. Pole and place of electrojet engines in the Russian Space Program (*in Russian*) / Trudy MAI (electronic journal). Issue 60. https:// www.mai.ru/science/trudy/.
- 7 Vasin A. I., Koroteev A. S., Lovtsov A. S. Overview of the work on electrojet engines at the Keldysh Research Center (*in Russian*). Trudy MAI (electronic journal). Issue 60. https://www.mai.ru/science/trudy/
- 8 Kim V. Stationary plasma engines in Russia: problems and prospects (in Russian). Trudy MAI (electronic journal). Issue 60. https:// www.mai.ru/science/trudy/
- 9 Gorshkov O. A., Muravlev V. A., Shagaida A. A. Hall and Ion Plasma Engines for Spacecraft (in Russian). Moscow: Mashinostroyeniye, 2008. 280 pp.
- 10 Narimanov G. S. Basics of Spacecraft Flight and Design Theory (in Russian). Moscow.: Mashinostroyeniye, 1972. 607 pp.
- 11 Petukhov V. G. Robust suboptimal feedback control for low-trust transfer between noncoplanar elliptical and circular orbits (*in Russian*). Spacecraft and Rockets. Vestnik Moskovskogo Aviatsionnogo Instituta. 2010. V. 17, No. 3. Pp. 50 58.
- 12 Lebedev V. N. Low-Thrust Spacecraft Flight Calculation (in Russian). Moscow: Coputing Center of the USSR Academy of Sciences, 1968. 105 pp.
- 13 Grishin S. D., Zakharov Yu. A., Odolevsky V. K. Low-Thrust Spacecraft Design (in Russian). Moscow: Mashinostroyeniye, 1990. 223 pp.
- 14 Ilyin V. A., Kuzmak G. E. Optimal Spacecraft Orbit Transfers (in Russian). Moscow: Nauka, 1976. 740 pp.
- 15 Grozdovsky G. L., Ivanov Yu. N., Tokarev V. V. Space Flight Mechanics (Optimization Problems) (in Russian). Moscow: Nauka, 1975. 704 pp.
- 16 Salmin V. V. Methods for the Solution of Variational Problems in Low-Thrust Space Flight Mechanics (*in Russian*). Samara : Samara Research Center of the Russian Academy of Sciences, 2006. 127pp.
- 17 Petukhov V. G. Optimization of interplanetary trajectories for spacecraft with ideally regulated engines using the continuation method. Cosmic Research. 2008. V. 4. No. 3. Pp. 219–232.

18 Grigoriev I. S., Zapletin M. P. Choosing promising sequences of asteroids. Automation and Remote Control. 2013. V. 74. Issue 8. Pp. 1284 – 1296.
19 *Papadimitriou C., Steiglitz K.* Combinatory Optimization. Algorithms and Complexity (*in Russian*) / Moscow:

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<sup>Mir, 1982. – 510 pp.
20 Sigal I. Kh. Algorithms for solving the two-criterion large-scale travelling salesman problems. Computational Mathematics and Mathematical Physics. 1994. V. 34. No. 1. Pp. 33–43.</sup>