

V. G. KOMAROV

OPTIMAL TRAJECTORIES OF SPACECRAFT UNDERGOING DISCONTINUOUS CHANGES IN ITS PARAMETERS

The work objective is to examine optimal coplanar trajectories of the spacecraft with parameters undergoing discontinuous changes within a certain time in the central-force Newton field.

The Pontryagin maximum principle for cases of discontinuous changes in parameters and coordinates is used as a research method.

Desired conditions of an optimal spacecraft control considering discontinuity are found.

The results can be employed to design trajectories of the near space flight of multistage rockets, as well as trajectories for the moon-planetary flights.

Keywords: *spacecraft, discontinuity, optimality, control, parameter, mass, time, motion, orbit, transition, trajectory, maximum principle.*

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