

EXPERIMENT-AND-CALCULATION DETERMINATION OF THE COEFFICIENTS APPEARING IN A MATHEMATICAL MODEL OF CAVITATING PUMPS OF LIQUID-PROPELLANT ROCKET ENGINES

Institute of Technical Mechanics

*of the National Academy of Sciences of Ukraine and the State Space Agency of Ukraine
15 Ieshko-Popel St., Dnipro 49005, Ukraine; e-mail: dolmrut@gmail.com*

Cavitation phenomena in liquid-propellant rocket engine (LPRE) pumps not only affect the power performance characteristics of the pumps, but they also affect the pump dynamics and pogo vibrations. The theoretical characterization of cavitation phenomena in LPRE pumps is not a widely used practice because theoretical and experimental data are in unsatisfactory agreement. Because of this, use is made of approaches that employ experimental data. The goal of this work is to determine the coefficients of a hydrodynamic model of cavitating LPRE pumps throughout the cavity existence region based on the experimental frequencies of cavitation oscillations and cavitation self-oscillation boundaries. In determining the cavity elasticity and negative resistance, use was made of the experimental cavitation oscillation frequencies of 26 LPRE pumps differing in dimensions and capacity. In determining the cavitation resistance distribution coefficient and the cavity-due disturbance transfer time, the experimental cavitation self-oscillation boundaries of 14 more pumps were used. To extend the cavity elasticity determination region, the extrapolation dependence of the cavity elasticity in cavitation stall regimes was updated. To make the stratification of the cavity resistance dependence more uniform in the range of large discharge coefficients, incipient cavitation numbers were refined. Using the qualitative dependence of the cavitation resistance distribution coefficient obtained from theoretical transfer matrices of cavitating pumps and its lower estimate (at zero disturbance transfer time) and upper estimate (for a uniform stratification of pump transfer matrix determinants), its analytical dependence was found. Using it and the coefficients of a mathematical model of cavitation oscillations on the cavitation-self oscillation boundary, disturbance transfer times were found and approximated.

Keywords: *liquid-propellant rocket engine, inducer-equipped centrifugal pump, cavitation, hydrodynamic model, experimental cavitation self-oscillation frequencies and regions.*

1. Borovsky B. I., Ershov N. S., Ovsyannikov B. V., Petrov V. I., Chebaevsky V. F., Shapiro A. S. High-Speed Blade Pumps. Moscow: Mashinostroyeniye, 1975. 336 pp. (in Russian).
2. Shevyakov A. A., Kalnin V. M., Naumenkova M. V., Dyatlov V. G. Theory of Automatic Rocket Engine Control. Moscow: Mashinostroyeniye, 1978. 288 pp. (in Russian).
3. Natanzon M. S. Pogo Vibrations of a Liquid-Propellant Rocket. Moscow: Mashinostroyeniye, 1977. 208 pp. (in Russian).
4. Pylypenko O. V., Degtyarev M. A., Nikolayev O. D., Klimenko D. V., DolgopoloV S. I., Khoriak N. V., Bashliy I. D., Silkin L. A. Providing of POGO stability of the Cyclone-4M launch vehicle. Space Sci. & Technol. 2020. V. 26. No. 4. Pp. 3-20.
<https://doi.org/10.15407/knit2020.04.003>
5. Ng S. L., Brennen C. E. Experiments on the dynamic behavior of cavitating pumps. ASME J. Fluids Eng. 1978. V. 100. Pp. 166-176.
<https://doi.org/10.1115/1.3448625>
6. C. E. Brennen, C. Meissner, E. Y. Lo, G. S. Hoffman. Scale effects in the dynamic transfer functions for cavitating inducers. ASME J. Fluids Eng., 1982. V. 104. Pp. 428-433.
<https://doi.org/10.1115/1.3241875>
7. Pilipenko V. V. Experiment-and-calculation method for determining the elasticity and volume of cavities in inducer-equipped centrifugal pumps. Izvestiya AN SSSR. Energetika i Transport. 1976.No. 3. Pp. 131-139. (in Russian).
8. Grigor'ev Yu. E., Pilipenko V. V. Experiment-and-calculation determination of cavity elasticity in inducer-equipped centrifugal pumps in the presence of backflows. In: Pump System Dynamics. Kiev: Naukova Dumka, 1980. Pp. 37-46. (in Russian).

9. Pilipenko V. V., Dolgoplov S. I. Experiment-and-calculation determination of the coefficients appearing in a vavity dynamics equation for inducer-equipped centrifugal pumps of different standard sizes. *Teh. Meh.* 1998. No. 8. Pp. 50-56. (in Russian).
[https://doi.org/10.1016/S0262-1762\(99\)80457-X](https://doi.org/10.1016/S0262-1762(99)80457-X)
10. Pilipenko V. V., Dovgot'ko N. I., Dolgoplov, S.I., Nikolaev, A.D., Serenko, V.A., Khoryak, N.V. Theoretical evaluation of the amplitudes of pogo vibrations in liquid propellant launch vehicles. *Kosm. Nauka Tsnol.* 1999. V. 5. No. 1. Pp. 90-96. (in Russian).
<https://doi.org/10.15407/knit1999.01.090>
11. Pilipenko V. V., Dovgot'ko N. I., Nikolaev A. D., Dolgoplov S. I., Serenko V. A., Khoryak N. V. Theoretical determination of dynamic loads (longitudinal vibration accelerations) on the RS-20 liquid-propellant rocket structure in its active flight. *Teh. Meh.* 2000. No. 1. Pp. 3-18. (in Russian).
12. Pylypenko O. V., Prokopchuk A. A., Dolgoplov S. I., Khoryak N. V., Nikolaev A. D., Pisarenko V. Yu., Kovalenko V. N. Mathematical simulation and stability analysis of low-frequency processes in a liquid-propellant staged-combustion sustainer rocket engine. *Vestnik Dvigatelistroyeniya.* 2017. No. 2. Pp. 34-42. (in Russian).
13. Pylypenko O. V., Prokopchuk O. O., Dolgoplov S. I., Nikolayev O. D., Khoriak N. V., Pysarenko V. Yu., Bashliy I. D., Polskykh S. V. Mathematical modeling of start-up transients at clustered propulsion system with POGO-suppressors for Cyclon-4M launch vehicle. *Space Sci. & Technol.* 2021. V. 27. No. 6. Pp. 3-15.
<https://doi.org/10.15407/knit2021.06.003>
14. Dolgoplov S. I., Nikolayev O. D., Khoriak N. V. Dynamic interaction between clustered liquid propellant rocket engines under their asynchronous start-ups. *Propulsion and Power Research.* 2021. V. 10. No. 4. Pp. 347-359.
<https://doi.org/10.1016/j.jprr.2021.12.001>
15. Pylypenko O. V., Dolgoplov S. I., Khoriak N. V., Nikolayev O. D. Procedure for determining the effect of internal and external factors on the startup thrust spread of a liquid-propellant rocket engine. *Teh. Meh.* 2021. No. 4. Pp. 7-17. (in Ukrainian).
<https://doi.org/10.15407/itm2021.04.007>
16. Koptilyy D., Marchan R., Dolgoplov S., Nikolayev O. Mathematical modeling of transient processes during start-up of main liquid propellant engine under hot test conditions. 8th European Conference for Aeronautics and Space Sciences (1-4 July, Madrid). 2019. 15 pp.
17. Dolgoplov S. I. Verification of a hydrodynamic model of a liquid-propellant rocket engine's cavitating pumps using experimental and theoretical pump transfer matrices. *Teh. Meh.* 2020. No. 3. Pp. 18-29. (in Ukrainian).
<https://doi.org/10.15407/itm2020.03.018>
18. Pilipenko V. V., Zadontsev V. A., Natanzon M. S. *Cavitation Oscillations and Hydrosystem Dynamics.* Moscow: Mashinostroyeniye, 1977. 352 pp. (in Russian).
19. Dolgoplov S. I. Generalized experimental-calculated liquid inertia resistance coefficient due to backflows at the inducer-equipped centrifugal pump inlet. *Teh. Meh.* 1995. No. 4. Pp. 99-103. (in Russian).
20. Bronshtein I. N., Semendyaev K. A. *Handbook on Mathematics for Engineers and Students.* Moscow: Nauka, 1980. 976 pp. (in Russian).
21. Dolgoplov S. I. Generalization of experimental elasticity of cavitation bubbles in LRE pumps that differ significantly in size and performance. *Sci. Innov.* 2023. V. 19, No. 5. Pp. 71-88.
22. Zadontsev V.A. Experimental study of LR pump at cavitation autooscillations regimes. *Proceedings of Third China-Russia-Ukraine Symposium on Astronautical Science and Technology, XI AN China.* (16-20 September). 1994. Pp. 285-287.
23. Zadontsev V. A., Drozd V. A., Dolgoplov S. I., Grabovskaya T. A. Off-line dynamic testing of a large LPRE

inducer-equipped centrifugal pump under cavitation self-oscillations. *Aviatsionno-Kosmicheskaya Tekhnika i Tekhnologia*. 2009. No. 9. Pp. 100-106. (in Russian).

24. Selofonov V. S. Study of LPRE Pump Dynamics in Cavitation. Ph.D. Thesis. Moscow, 1972. 229 pp. (in Russian).
25. Zadontsev V. A., Drozd V. A., Dolgopolov S. I., Grabovskaya T. A. Off-line testing of the Zenith launch vehicle second-stage sustainer engine's oxidizer pump under cavitation self-oscillations. *Aviatsionno-Kosmicheskaya Tekhnika i Tekhnologia*. 2010. No. 10. Pp. 89-93. (in Russian).
26. Ivanov Ya. N. Experimental study into efficient dampers of cavitation self-oscillations in an LPRE propellant feed system. *Vestnik of S. P. Korolev Samara State Aerospace University*. 2006. No. 2. Pp. 357-360. (in Russian).
27. Zhulai Yu. A. Dynamic testing of an inducer-equipped centrifugal pump under cavitation oscillations. *Vestnik Dvigatelaystryeniya*. 2006. No. 3. Pp. 141-145. (in Russian).
28. Ershov N. S. Experimental study of pump system cavitation oscillations. *Pump System Dynamics*. 1980. Pp. 3-9. (in Russian).
29. Dovgot'ko N. I. On a study of inducer-equipped centrifugal pump - pipelines system stability for cavitation self-oscillations. *Pump System Dynamics*. 1980. Pp. 9-14. (in Russian).
30. Drozd V. A., Zadontsev V. A., Khodursky V. E. Experimental determination of the eigenfrequency and decrement of liquid oscillations in a feed line - LPRE pump system. *Tekhnicheskaya Mekhanika Raketno-Kosmicheskikh Sistem*. 1986. Iss. 1. Pp. 90-96. (in Russian).
31. Natanzon M. S., Bal'tsev N. I., Bazhanov V. V., Leidervarger M. R. Experimental study of cavitation oscillations in an inducer-equipped centrifugal pump. *Izvestiya AN SSSR. Energetika i Transport*. 1973. No. 2. Pp. 151-157. (in Russian).
32. Shakutina L. G. Effect of partial cavitation in an inducer on pump and LPRE dynamic performance in the low-frequency range. Ph.D. Thesis. 1971. 166 pp. (in Russian)
33. Dolgopolov S. I. Generalization of experimental stall pressures in cavitating LPRE inducer-equipped centrifugal pumps. *Space Technology. Missile Armaments*. 2007. Iss. 1. Pp. 98-108. (in Russian).
34. Dolgopolov S. I. Determining the coefficients of a hydrodynamic model of cavitating pumps of liquid-propellant rocket engines from their theoretical transfer matrices. *Teh. Meh.* 2024. No. 1. Pp. 3-12. (in Ukrainian).
<https://doi.org/10.15407/itm2024.01.016>

Received on July 8, 2024,
in final form on September 30, 2024